

3503, H35

## 35 Series

— 58 or 70 HP



2-Stroke

Highest power to weight ratio

 Perfect for light aviation, hovercraft, gyro or helicopters



DESCRIPTION

The 35 Series liquid cooled two-stroke engines offer the maximum power-to-weight ratios available in the 70-horsepower market.

Offers an ideal choice for use in light and ultralight aviation, hovercraft, gyro and helicopters and all applications where weight could be an issue.

Offers a time between overhauls of 1000 hours and is well known for its ease of maintenance and exceptional durability.



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## 35 Series

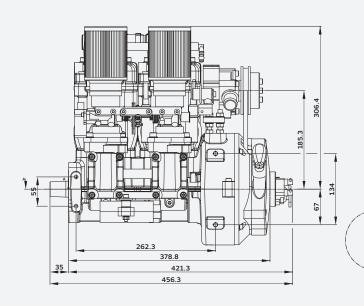


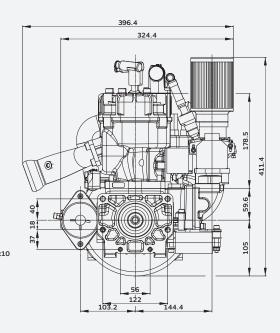
## TECHNICAL SPECIFICATION:

TYPE:	Two cylinder two stroke (inline)
DISPLACEMENT:	<b>625 cm³</b> (38,1 in³)
STROKE:	<b>69 mm</b> (2,72 in)
BORE:	<b>76 mm</b> (2,99 in)
MAX. PERFORMANCE:	<b>52 kW (70 HP)</b> at 6500 rpm (3503) <b>43 kW (58 HP)</b> at 5200 rpm (H35) According DIN 70020
MAX. TORQUE:	<b>77,3 Nm (57 ft.lb)</b> at 6000 rpm (3503) <b>79 Nm (58 ft.lb)</b> at 5000 rpm (H35)
CARBURATION:	multi point injection (E) or 2x carburetor (V)

Mixture 1:50, 2-stroke-oil, fuel min. 95 octane (RON) Mixture 1:80-100 with BLUEMAX 2-stroke-oil, fuel min. 95 octane

WEIGHT:	36 kg (79,0 lb) with exhaust and water in the engine
LENGTH:	<b>472 mm</b> (18.58 in)
WIDTH:	<b>410 mm</b> (16.14 in)
HEIGHT:	<b>360 mm</b> (14.17 in)
STARTING DEVICE:	Recoil starter
RUNNING DIRECTION:	Counter-clockwise, view to output shaft
COOLING:	Liquid cooling
IGNITION SYSTEM:	CDI





## OPTIONS

**FUEL MIXTURE:** 

- Dual ignition
- **Gear box G 50** (1:2,16/1:2,29/1:2,59/1:3,16/1:3,65)
- Electric starter
- Separate lubrication

This is not a certificated aircraft engine! It has not received the safety and durability testings specified by aircraft standards. It is only for use in uncertificated experimental aircraft or vehicles when there is no risk for the safety due to an engine failure. Never fly the aircraft equipped with this engine in circumstances or in areas, in weather-conditions or in altitudes where you have no chance for successful landing after an engine failure. The user is taking all risk resulting from the use of this engine and he is aware of the possibility of sudden functional disturbances.

